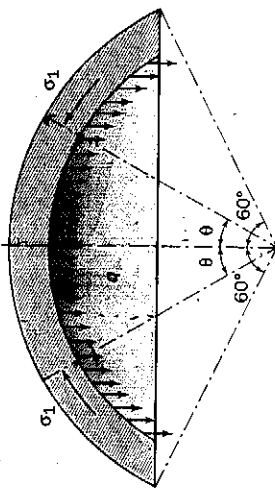


Comparing eqns. [2.10] and [2.11] it is seen that the circumferential stress is twice the axial stress. Figure 2.12(d) shows a small element of the shell subjected to the axial and circumferential (hoop) stresses.

Example 2.2

A concrete dome is 250 mm thick, has a radius of 30 m and subtends an angle of 120° at the support ring. Calculate the stresses at the supports due to self-weight. The density for concrete is 2.3 Mg/m^3 . The dome is illustrated in Fig. 2.13.

Fig. 2.13



$$F = \sigma \cdot A$$

$$\text{radial component} = \sigma_1 \sin \theta$$

Consider the vertical equilibrium of a circular segment containing an angle 2θ ; then if q is the force per unit area due to weight of concrete,

$$(2\pi r \sin \theta \times t \sigma_1 \sin \theta) + 2\pi r(r - r \cos \theta)q = 0$$

$$\sigma_1 = -\frac{qr(1 - \cos \theta)}{t \sin^2 \theta} = -\frac{qr}{t} \frac{1}{(1 + \cos \theta)}$$

Since self-weight, unlike applied pressure, does not act radially everywhere, eqn. [2.8] has to be modified to take account of the radially resolved component of weight, giving

$$\frac{\sigma_1}{r_1} + \frac{\sigma_2}{r_2} = -\frac{q}{t} \cos \theta$$

Substituting for σ_1 ,

$$\sigma_2 = \frac{qr}{t} \left(\frac{1}{1 + \cos \theta} - \cos \theta \right)$$

The stresses at the supports are obtained by putting $\theta = 60^\circ$ and $q = 2.3 \times 9.81 \times 250/1000 = 5.65 \text{ kN/m}^2$:

$$\sigma_1 = -\frac{5.65 \times 30}{0.25} \times \frac{1}{1^{\frac{1}{2}}} = -452 \text{ kN/m}^2$$

$$\sigma_2 = \frac{5.65 \times 30}{0.25} \left(\frac{2}{3} - \frac{1}{2} \right) = 113 \text{ kN/m}^2$$